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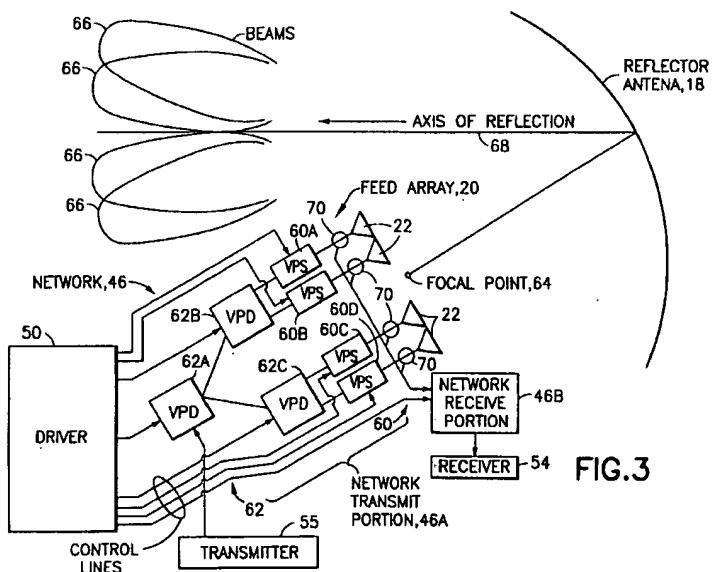
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(54) Positionable satellite antenna with reconfigurable beam

(57) An antenna system (10A), is suitable for use on a communication satellite encircling the earth, and has a reflector (18) illuminated by a set of feed elements (22) wherein one of the feed elements generates a primary beam (30). The reflector (18) is shaped to establish a specific configuration of the beam, and, upon illumination of the earth, there results a specific shape to the beam footprint (38). A mechanical positioning device (16) connects the antenna with a body (14) of the space-craft for adjusting an orientation of the feed antenna (12)

and its beam (30) relative to the body (14) of the space-craft. Control circuitry (26) for adjustment of signal strengths and phase shifts of signals sent to respective ones of the feeds (22) is accomplished by electronic circuitry including a memory (52) which stores coefficients employed in the operation of variable power dividers (62) and variable phase shifters (60) of a feed network (46A) supplying electromagnetic signals to the feed. The memory is addressed to provide a desired correction to beam configuration corresponding to a specific orientation of the antenna.



Description

[0001] This invention relates to mechanically steerable antennas, suitable for use on board a communications satellite encircling the earth and, more particularly, to an antenna having an array of feed elements illuminating a reflector wherein the reflector is a shaped-beam reflector and wherein, upon a repositioning of the antenna relative to the satellite, there is an adjustment of the beam configuration.

[0002] In the use of communication satellites, one or more antennas on board the satellite direct beams of radiation to selected locations on the earth's surface as the satellite progresses in a stationary orbit about the earth. Based on considerations of communication traffic, it may be desirable to reorient an antenna relative to the body of the satellite so as to illuminate another portion of the earth's surface. Each portion of the earth's surface to be illuminated is characterized by a desired beam footprint designating the specific area of the earth's surface wherein are located receiving and/or transmitting stations which are to communicate with the satellite via the antenna. The antenna includes both a reflector and an array of feeds which illuminate the reflector to produce a beam configuration and corresponding footprint. However, with a repositioning of the antenna to illuminate different regions of the earth, it is desirable frequently to adjust the configuration of the footprint to meet local traffic conditions.

[0003] By way of example, two antennas on board a single satellite may be directed to illuminate various areas of the earth's surface in footprints configured to overlap edge regions of neighbouring footprints. Or, antennas carried by different satellites may cooperate by illuminating various areas of the earth's surface wherein footprints from the beams of the various antennas are to overlap slightly at the peripheral regions of the respective beam footprints. The configurations of the various footprints may vary, depending on the viewing angles by which the satellite directs the beam to the earth. The footprints may vary such that a relatively small populated region of the earth may be assigned to one beam. A much larger region of the earth of relatively sparse population may employ a single antenna covering the larger region. Therefore, upon a repositioning of a beam, it may be necessary to provide for adjustment of the beam configuration.

[0004] It is desirable to be able to adjust the footprint in a generally continuous fashion so as to allow for various configurations of footprints which may become desirable as a result of shifting patterns of population on the earth and shifting demands for the satellite service. Prior communication antennas have been inadequate in this respect in that their capacity for alteration of footprint configuration was limited to a relatively small number of footprints obtainable by switching various feeds of a feed network.

[0005] A problem exists in that presently available

communication antennas while being mechanically steerable do not provide a desired capacity for precise adjustment of footprint configuration.

[0006] The present invention seeks to improve the capability for adjustment of footprint configuration by means of an antenna system suitable for being carried by a satellite.

[0007] According to the invention there is provided an antenna system suitable for being carried on board a satellite travelling about the earth, the antenna system including an antenna comprising a reflector and an array of feed elements which illuminate the reflector to produce a beam of radiation emitted by the antenna, the system comprising positioning means operative to orient the antenna relative to a body of the satellite for illuminating a desired position of the earth with the beam, a beam controller having a feed network for applying signals to respective ones of the feed elements, wherein the reflector is a shaped reflector operative with the feed elements to produce a beam footprint, and the beam controller is operative to adjust the configuration of the beam to reconfigure the beam footprint to provide a desired footprint on the surface of the earth and the beam controller adjusts relative amplitudes and phases of signals fed to respective ones of the feed elements.

[0008] The system may include an antenna which comprises a reflector illuminated by an array of feed elements, the array being much smaller than a diameter of the reflector. Typically, the reflector has the general shape of a section of an ellipsoid or parabola, by way of example, and the reflector is further shaped to provide for a specific configuration of beam. Upon deployment of a satellite encircling the earth, the beam configuration produces a footprint on the surface of the earth.

[0009] An antenna positioning mechanism is provided for moving the antenna relative to a body of the spacecraft. A repositioning of the beam results in illumination of a different portion of the earth's surface with a possible consequential need for an adjustment of the configuration of the beam footprint upon the earth's surface. The invention provides for adjustment of the configuration of the beam configuration. Such adjustment may be by adjustment of relative signal strengths and for phase shifts among signals of elements of the feed. Due to the

relatively small size of the array of feed elements, relative to the diameter of the reflector, the primary contribution to the configuration of the beam is the geometry of the reflector. An offsetting in the position of one feed element relative to another feed element results in an inclination of their respective beams relative to each other. Summation of the electric and magnetic fields of the respective beams produces a resultant beam having a footprint different from the footprint of any one of the beams. Adjustment of the relative phases and/or amplitudes of the signals of the respective feed elements results in an adjustment of the configuration of the resultant beam to suit a specific situation to be handled by the communication satellite.

[0010] Beam control circuitry suitable for adjustment of the beam configuration typically comprises a set of adjustable power dividers and adjustable phase shifters connected in a network for applying the desired signals to respective ones of the elements of the feed. Adjustment occurs by multiple steps to approximate a continuously variable (analog) form of adjustment. Operation of the power dividers and of the phase shifters may be controlled electronically by means of digital coefficients stored in a memory wherein specific sets of coefficients have been previously established for development of a specific beam footprint. New footprint configurations can be developed by sensing beam intensity at various locations on the earth's surface and calculating, via simulation on a computer, the necessary coefficients for the new footprint. These coefficients can then be transmitted to the satellite for storage in the memory.

[0011] In order that the invention and its various other preferred features may be understood more easily, some embodiments thereof will now be described, by way of example only, with reference to the drawings, in which:-

Fig. 1 is a stylized view of a satellite encircling the earth and carrying an antenna constructed in accordance with the invention,

Fig. 2 is a block diagram showing components of the antenna and control circuitry for operation of the antenna, and

Fig. 3 shows circuitry for driving the elements of the antenna.

[0012] Identically labelled elements appearing in different ones of the figures refer to the same element but may not be referenced in the description for all figures. [0013] Fig. 1 shows a satellite 10 carrying an antenna system 10A which includes an antenna 12 affixed to a body 14 of the satellite 10 by a positioning mechanism 16. The positioning mechanism 16 serves to pivot the antenna 12 relative to the body 14 to provide for an orientation of the antenna 12 in a desired direction relative to the satellite 10. The antenna 12 comprises a reflector 18 and an array 20 of feed elements 22 which illuminate a front concave side of the reflector 18. A frame 24 extends from a base portion of the reflector 18 for supporting the feed elements 22 at a location at or near a focus of the reflector 18 and offset from an axis of a main beam of the reflector 18. The satellite 10 is constructed in conventional fashion, and includes electronic circuitry 26 powered with electric power provided by solar panels 28 carried by the satellite 10. The electronic circuitry 26 serves to generate electromagnetic signals radiated as a beam 30 from the antenna 12, and to control a cross-sectional shape of the beam 30. Alternative positions of the beam 30 are indicated at 32 and 34, which alternative positions result from a pivoting of the antenna 12 by

the positioning mechanism 16.

[0014] The satellite 10 travels in an orbit about the earth 36 with the antenna 12 facing the earth for illumination of a portion of the earth's surface by the beam 30. In the event that the antenna 12 has been pivoted, illumination of another portion of the earth's surface is accomplished via the beam 32 or the beam 34, by way of example. In the illumination of a portion of the earth's surface by the beam 30, an actual footprint 38 of an illumination pattern is shown in solid line while a desired configuration of the footprint is indicated by a dashed line at 40. Various ground stations 42 are shown at different locations on the earth's surface for receiving radiated signals (down-link) transmitted from the satellite 10, and for transmitting signals (up-link) to the satellite 10. One of the ground stations 42, namely station 42A may be provided with a computer 44 for calculation of a desired set of coefficients for control of beam cross section, in accordance with the invention, as will be described hereinafter.

[0015] With reference to Fig. 2, the electronics unit 26 comprises a feed network 46 connecting with an assembly 48 of the feed elements 22 for imparting phase and amplitude to each of the respective signals that are applied to various ones of the feed elements 22. A driver 50 is responsive to coefficients of phase and amplitude stored within a memory 52 for driving variable power dividers and variable phase shifters of the feed network 46 (as will be described in Fig. 3) to impart the desired relative amplitudes and relative phases among the signals of the various feed elements 22. Also included within the electronics circuitry 26 are a transmit/receive unit 53 which includes a receiver 54 of up-link signals and a transmitter 55 of down-link signals, a command and control receiver 56, and a communications control unit 58. The receivers 54 and 56 and the transmitter 55 connect with the feed assembly 48 via the feed network 46. The control unit 58 connects with the command and control receiver 56 and the transmit/receive unit 53.

[0016] In the operation of the satellite 10 to provide the communications function, the feed network 46 is operative to pass both up-link signals from the feed assembly 48 to the transceiver 54 and down-link signals from the transceiver 54 to the feed assembly 48. The feed network 46 imparts various phases and amplitudes to signals of respective ones of the feed elements 22, in a manner to be described with reference to Fig. 3, to provide a desired cross-sectional configuration to a beam, such as the beam 30 (Fig. 1) produced by the antenna 12. The transmit/receive unit 53 is operative in response to signals of the communications control unit 58 for receiving up-link signals in various up-link channels and for retransmitting the signals as down-link signals in various down-link channels. Included within an up-link signal transmitted from the ground station 42A (Fig. 1) are command and control signals to the receiver 56 for operation of the positioning mechanism 16 to direct a specific orientation of the antenna 12, for selecting coeffi-

clients of the memory 52 to configure a beam such as the beam 30, and for instructing the communications control unit 58 in the handling of communications traffic. [0017] Fig. 3 shows details in the construction of the antenna 12 including also details in the construction of the feed network 46 connecting with the feed elements 22 of the antenna 12, four of the feed elements 22 being shown by way of example. The feed network 46 comprises a transmit portion 46A and a receive portion 46B. The network transmit portion 46A comprises a plurality of variable phase shifters (VPS) 60 of which individual ones are further identified as 60A, 60B, 60C and 60D, and a plurality of variable power dividers (VPD) 62 of which individual ones thereof are further identified as 62A, 62B and 62C. The array 20 of the feed elements 22 is positioned, preferably, at a focal point 64 of the reflector 18. Unlike a phased-array antenna wherein the radiations of numerous individual feed elements are summed together with specified phase and amplitude tapers to produce a beam in a desired direction, the reflector 18 serves the function of gathering and collimating the rays of radiation emitted by respective ones of the feed elements 22 to produce desired beams of radiation. Each of the respective feed elements 22 also acts with the reflector 18 to produce a beam of radiation, four such beams being shown, by way of example, as beams 66 disposed symmetrically about an axis 68 of an array of the beams.

[0018] The network receive portion 46B is constructed in a fashion analogous to the construction of the network transmit portion 46A, and may include phase shifters (not shown) for setting relative phases among received signals of the feed elements 22, and power combiners (not shown) for combining received signals of the various feed elements 22 to form the received beam. In order to separate a transmitted signal from a received signal for each of the respective feed elements 22, signal separators 70 interconnect respective ones of the feed elements 22 with both the transmit portion 46A and the receive portion 46B of the network 46. By way of example, a signal separator 70 may be an orthogonal mode junction (OMJ) for the case of orthogonally polarized transmit and received signals, or a diplexer in the case of copolarized signals.

[0019] The electric and the magnetic fields of respective ones of the beams 66 sum together in the far field of the reflector 18 to produce a beam, such as the beam 30 in Fig. 1, having a desired cross-sectional configuration. For forming the beam 30, the radiating aperture of the reflector 18 is greater than the radiating aperture of the feed array 20, at least by a factor of approximately ten for improved directivity of the beam 30, a factor in the range of 50-100 being used in a preferred embodiment of the invention. Adjustment of relative phases and amplitude of the signals emitted by each of the feed elements 22, with respect to each other effects the summation of the electric and magnetic fields of the beams 66 for control of the cross-sectional configuration of the

resulting beam 30.

[0020] The operation of the feed network 46 may be described with respect to the transmission of a down-link signal via the network transmit portion 46A. In the 5 transmission of a down-link signal, a transmitter 55 outputs the down-link signal to power divider 62A which operates to divide the power of the signal between power dividers 62B and 62C. The power divider 62B provides for a further division of the signal power among two signal 10 channels, and the power divider 62C operates similarly to divide its signal power among two further signal channels. Each of the signal channels comprises a phase shifter 60 and a feed element 22. Power for the first signal channel is provided by the power divider 62B in concert with the phase shifter 60A, power for the second signal channel is provided by the power divider 62B in concert with the phase shifter 60B, power for the third signal channel is provided by the power divider 62C in concert with the phase shifter 60C, and power for the 15 fourth signal channel is provided by the power divider 62C in concert with the phase shifter 60D.

[0021] Control signal lines are provided for respective 20 ones of the power dividers 62 and the phase shifters 60 of the network transmit portion 46A, and also for the variable phase shifters and the variable power combiners (not shown) of the network receive portion 46B. The control signal lines connect with the driver 50 to enable the driver 50 to communicate, in the case of transmitted signals, with the respective power dividers 62 and phase 25 shifters 60 for commanding the respective power dividers 62 and phase shifters 60 to provide, respectively, desired power divisions among the signal channels. Corresponding operation of the driver applies for control of the components of the network receive portion 46B for the case of received signals.

[0022] With respect to the operation of the antenna system 10 (Figs. 1-3), the command and control receiver 56 supplies the phase and amplitude coefficients to the memory 52 in accordance with instructions received 30 from the ground station 42A. The memory 52 is also provided with coefficients of phase and amplitude prior to a launching of the satellite 10, which coefficients serve to define a beam configuration in the absence of specific requests by the ground station 42A. This allows ground 35 control to establish the setting of any of the variable components of the network 46 to obtain any of numerous beam shapes. Thus, the memory 52, in conjunction with the driver 50 and the feed network 46, serves as a beam controller for controlling the configuration of the beam 30 radiated by the antenna 12. The command and control receiver 56 outputs two forms of signals, one form of signal being the coefficients to be stored in memory 52, and the other form of signal being an address for addressing the memory 52 to output a specific set of 40 the coefficients to the driver 50 for accomplishing the desired beam configuration.

[0023] The command and control receiver 56 also serves as a director of antenna orientation by command-

ing the positioning mechanism 16 to orient the antenna 12 with a desired orientation relative to the body 14 of the satellite 10. For any position of the antenna 12, the shaped reflector 18 focuses radiation from respective ones of the feeds 22 to produce beams which sum together to give a scanned beam 72 of desired cross-sectional configuration. The specific locations of the feeds 22 is a matter of choice in the design of the antenna 12 for performance of a specific mission but, in a typical situation, the feeds 22 would be spaced apart by one to two wavelengths of the radiation emitted by the antenna 12, and the radiating aperture of an individual feed element would be approximately one wavelength in diameter.

[0024] The number of the feeds 22 is also determined by the mission of the satellite and the amount of control desired over peripheral regions of the beam footprint on the earth's surface. Thus, for very fine adjustment of the footprint, by way of example, a larger number of the feeds 22 would be employed. The feeds 22 may be distributed uniformly about the focal point 64 or, if adjustments are to be made primarily in an east-west direction of the footprint, by way of example, more of the feeds 22 may be disposed horizontally than vertically in the assembly 48 (Fig. 2). Continuous adjustment of the phase shifters 60 may be attained by either analog or digital circuitry, as is well known in the construction of phase shifters. Similarly, control of power division of the power dividers 62 for continuous adjustment of power ratios may be attained either by analog or digital circuitry, such circuitry being well known for adjustment of power ratios. Thereby, the invention is able to redirect a beam of radiation while controlling the configuration of the beam cross section.

Claims

1. An antenna system (10A) suitable for being carried on board a satellite (10) travelling about the earth, the antenna system including an antenna (12) comprising a reflector (18) and an array (48) of feed elements (22) which illuminate the reflector to produce a beam (30) of radiation emitted by the antenna, the system comprising positioning means (16) operative to orient the antenna (12) relative to a body (14) of the satellite (10) for illuminating a desired position of the earth with the beam (30), a beam controller having a feed network (46) for applying signals to respective ones of the feed elements (22), wherein the reflector (18) is a shaped reflector operative with the feed elements (22) to produce a beam footprint (38), and the beam controller is operative to adjust the configuration of the beam (30) to reconfigure the beam footprint to provide a desired footprint (40) on the surface of the earth and the beam controller adjusts relative amplitudes and phases of signals fed to respective

ones of the feed elements (22).

2. An antenna system as claimed in Claim 1, wherein the reflector (18) has a radiating aperture and the array of feed elements (22) has a radiating aperture, the radiating aperture of the reflector (18) being greater than the radiating aperture of the feed array (22) by at least approximately a factor of 10.
- 10 3. An antenna system as claimed in Claim 1 or 2, wherein individual feed elements (22) of the feed array (48) serve for adjustment of a configuration of a cross section of the beam (30).
- 15 4. An antenna system as claimed in any one of the preceding claims, wherein power fed to any one of the feed elements (22) is adjustable relative to power fed to another of the feed elements to establish a configuration of a cross section of the beam (30).
- 20 5. An antenna system as claimed in any one of the preceding claims, wherein a feed network (46) of the beam controller comprises a power divider network (62) having a plurality of variable power dividers (62A, 62B, 62C) distributing power among respective ones of the feed elements (22) for adjusting signal strength among respective ones of the feed elements.
- 25 30 6. An antenna system as claimed in any one of the preceding claims, wherein a feed network (46) of the beam controller comprises a plurality of phase shifters (60) connected to respective ones of the feed elements (22) for adjusting phases of signals of the respective feed elements.
- 35 7. An antenna system as claimed in claim 5, comprising a memory (52) storing coefficients for directing division of power among the variable power dividers (62A, 62B, 62C), and for directing phase shifts imparted to signals of the respective feeds (22) by respective ones of the phase shifters (60).

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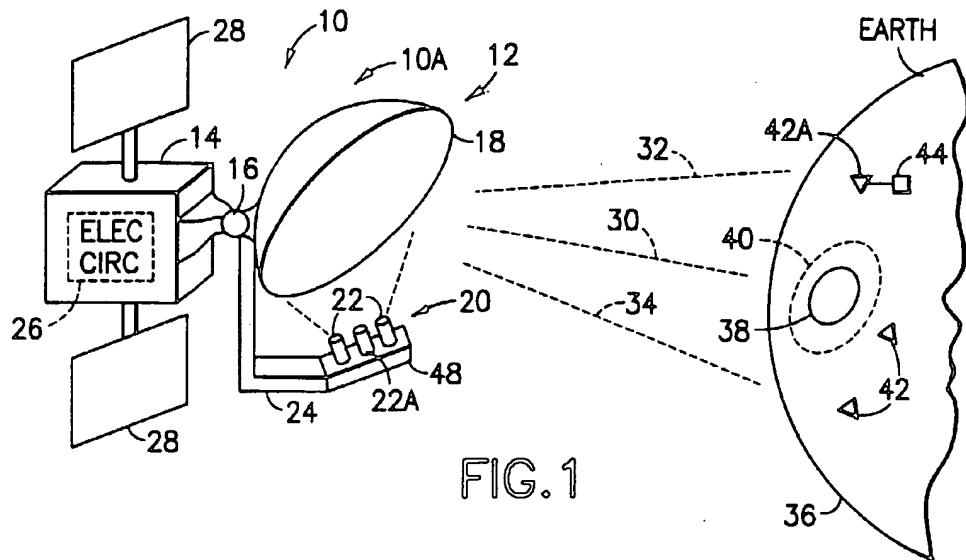


FIG. 1

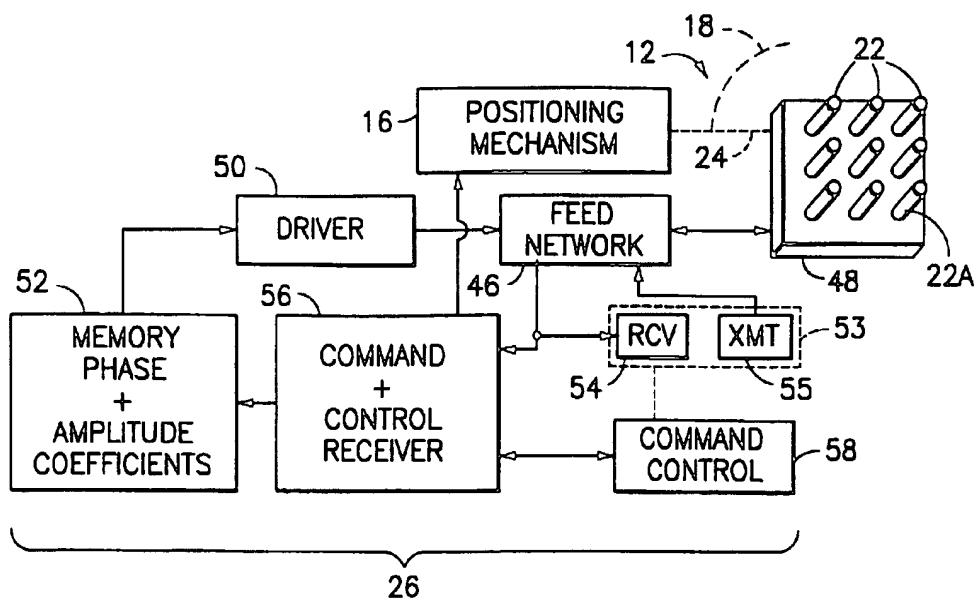


FIG. 2

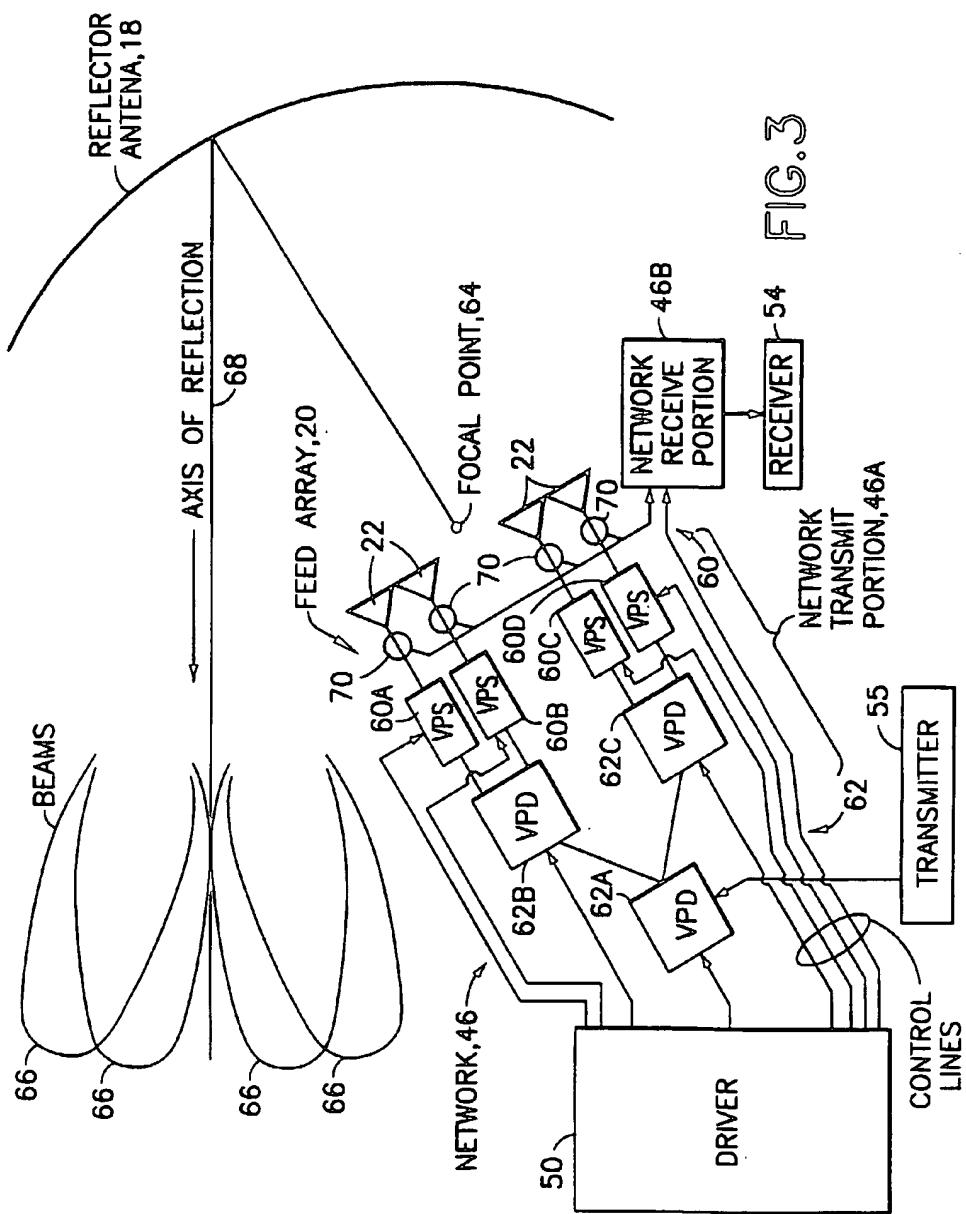


FIG. 3

RECEIVER 54

TRANSMITTER
62 55

NETWORK TRANSMIT PORTION, 46A

RECEIVER 54

NETWORK RECEIVE PORTION, 46B

FIG. 3



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EUROPEAN SEARCH REPORT

Application Number

EP 98 30 9113

DOCUMENTS CONSIDERED TO BE RELEVANT			CLASSIFICATION OF THE APPLICATION (Int.Cl.6)
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	
E	EP 0 845 834 A (LORAL SPACE SYSTEMS INC) 3 June 1998	1-4	H01Q3/26 H01Q19/10
P, X	* column 2, line 42-52; figure 1 * * column 1, line 3-27 *	1-4	H01Q1/28
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			TECHNICAL FIELDS SEARCHED (Int.Cl.6)
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<p>The present search report has been drawn up for all claims</p>			
Place of search	Date of completion of the search	Examiner	
MUNICH	12 March 1999	Felgel-Farnholz, W-D	
CATEGORY OF CITED DOCUMENTS		T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons S : member of the same patent family, corresponding document	
X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document			

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ON EUROPEAN PATENT APPLICATION NO.

EP 98 30 9113

This annex lists the patent family members relating to the patent documents cited in the above-mentioned European search report.
The members are as contained in the European Patent Office EDP file on
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